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A new methodology to optimize spiral bevel gear topography

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This paper aims to present the new method developed to generate optimized spiral bevel gear surfaces. Thanks to a complex non linear finite element model, the geometrical gear meshing positions under operational loads are first precisely computed. These meshing positions are then used as inputs of a calculation process that seeks to define the best tooth surface topography. So far, this activity was based on rigidity tests conducted directly by the designer, which led to repeat calculations whose progress was difficult to control. EUROCOPTER uses now optimization algorithms to compute automatically the surfaces of the tooth contact flanks. This approach leads to higher performances of the gear while reducing the development time. This paper describes the new process implemented to design the tooth shape, and illustrates its interest through an example.

1. Introduction regarding MGB specificities

Main Helicopter Gearboxes (MGB) are high-tech devices that must transmit very high mechanical power, with a significant reduction in speed, at lower weight while occupying the least space possible. So, it is necessary to undertake a heavy and long development process before reaching a gearbox design that satisfies all the economic, technical and security requirements and especially that allows thousands of flight hours [1]. Customers attach particular importance to weight criterion, this one being a key factor in terms of profitability. So, housings are generally made of light alloys like Aluminium or Magnesium. A large part of the rotor loads being transferred to the airframe by the MGB, especially very high torques, housing undergoes high deflections that make the design of MGB a specific case regarding gear optimization. Under flight loads, gears can suffer from a displacement in the order of magnitude of 1 mm. The gear specialist has to ensure a proper behaviour in a set of positions that are the results of MGB external loads conditions in conjunction with the MGB internal stiffness. An optimized mechanical behaviour is ensured mostly by a correct geometrical position of contact areas and an acceptable Hertz pressure. For that purpose, the designer has to optimize geometrically the initial tooth flank of the gears by a set of corrections between 10 and 200 \( \mu \)m. Finding the right optimization is sometimes very long and several loops of topography adjustments of a few microns may be required. The order of magnitude of the local deflection induced by Hertz pressure phenomena is no more than a few microns, so the optimization window is quite small. This sensitivity of the topography has also an impact on the manufacturing tolerances of power gears that are very narrow. The aim of this paper is to present a new methodology to optimize the topography of spiral bevel gears.

2. A review of previous works on spiral bevel gear simulation and optimization

2.1. Loaded meshing simulation

In 1987, Madrosky [2] proposes a loaded meshing model rather advanced for his time. Indeed, it already takes into account the flexibility of teeth. Those are calculated by the finite element method. The other part of the system is considered infinitely rigid. The relative displacements of the axes of the gears result from the deformation of the complete device. The contact stiffness is deduced from the Hertz theory. The latter gives contact dimensions and pressure. Gosselin [3] simulates the meshing of various tooth flank topographies, using the same techniques as Madrosky. He recommends maximizing contact ratio to improve load sharing. Lelkes [4] simulates the behaviour of gears manufactured according to the Pallod method, also referring to the same technique. However, the Hertz theory is based on assumptions inconsistent with some contact properties of spiral bevel gears. Indeed, the opposite surfaces are conforming, the bodies are finished and they slide in the contact zone. De Vaujany [5] opts for an approach that combines finite elements analyses to evaluate tooth bending effects and the Boussinesq and Cerruti theory to model the contact deformations. Alves [6] accelerates the computation of the stiffness of teeth, wheels and shafts by introducing a set of interpolation functions. Bibel [7] deals with the loaded meshing using directly PATRAN finite element models. The contact is treated by the computational code MARC. Litvin [8] and Argyris [9] adopt the same approach. However, they use the software Abaqus. The meshes are automatically built from the...
points generated on the tooth flanks without using a CAD software. The finite element modelling of the teeth is more accurate and faster. The work presented in this paper is based on a similar technique. The model proposed by Litvin [10] features three teeth (Fig. 1). The flexibility of the tooth is taken into account, the gear itself being considered fixed. A torque is applied along the pinion axis. The simulation results highlight every potential contact on tooth edges. Vimercati [11] also uses a computational code based on the finite element method. His work shows that the contact ratio increases with the torque while the quasi-static transmission error decreases.

Fig. 1. Model and boundary conditions [9]

2.2. Gear topography optimization

Papers dealing with the optimization of the loaded spiral bevel gear behaviour are relatively recent. Simon [12] analyses the influence of the tool geometry on the maximum contact pressure and the magnitude of the quasi-static transmission error. The considered tool profile is made of two tangent arcs. Their radius and the tool diameter are settings to be corrected. Simon calculates different cases of relative displacements of the teeth, as well as different intensities of the torque.

Artoni [13] proposes a fully automated optimization process. It consists of two steps. Firstly, an ideal «ease-off» is identified. Its shape must minimize the area that is not occupied by the contact path in a given zone. The latter is defined according to the ANSI/AGMA 2005-D03 standard. Artoni refers to the loaded meshing model developed by Bracci [14]. The optimization variables are the coefficients of the quartic function which define the «ease-off». Secondly, the appropriate settings of the machine tool are sought.

Artoni [15] then works on the minimization of the quasi-static transmission error. The maximum contact pressure is minimized simultaneously. The contact path must not touch a flank edge. Therefore, a constraint function requires zero contact pressure outside the area defined by the ANSI/AGMA 2005-D03 standard. The objective is the sum of the transmission error, pressure and constraint function. The optimization variables are bounded. The research of the optimal «ease-off» is based on the Hooke and Jeeves method. Artoni uses the loaded meshing model developed by Kolivand and Kahraman [16]. The results shown in Fig. 2 seem promising. However, the author does not specify whether they are obtained after determining the machine tool settings.

Fig. 2. Maximum contact pressure optimization [15]

Gabiccini [17] uses Artoni’s work [13]. His aim is the reduction of the sensitivity of the loaded contact path in order to bear the relative displacements of the gear axes. Tolerance ranges are assigned to the misalignment components. Every combination of their bounds is taken into account. The envelope of the displaced contact paths is plotted in Fig. 3.

Fig. 3. Envelope of the displaced contact paths [17]

3. Mastering the gearbox deflection

As explained in the introduction, the specificity of helicopter gearbox design is the need for gears optimization not in one position but in a set of positions. So, before computing tooth topography, the relative positions of all gear stages have to be computed precisely according to the various load configurations that can be found in flight. To estimate the gear displacements, EUROCOPTER has been working for years on the improvement of non linear mechanical finite element models that are now used as “Virtual Tests” [18]. Fig. 4 illustrates the kind of models that has been developed. These models simulate very well bearing behaviours [19] and give accurate load distributions and deflection on housings. A full meshing of all contact areas has then been introduced. To keep the computation time under control, specific meshing shapes of gear and bearing have been developed to minimize the number of contact elements. Even with those precautions the number of degrees of freedom can supersede easily several millions.

After a batch of simulations, the displacements of the tooth flanks can be synthesized into sets of four values: the axial displacements of the wheel and the pinion, the offset distance between the two gears and the variation angle between both axes. These parameters are found to be the most influential regarding spiral bevel gears optimization. Based on all these values coming from the FE analyses, the optimization phase of the tooth flank can be launched.

Fig. 4. Example of complex FEM models [18]

4. New optimization process

4.1. Contact pressure optimization process

The efficiency of a manual correction process depends on the operator experience. The latter cannot perceive clearly the links between the tooth flank shape and the transmission error. It is very difficult for him to establish the links between the changes that he implements on the shape and the implications that follow in terms of load sharing. Therefore, the search for a minimum contact pressure by setting himself the different designs to be evaluated is
The loaded contact simulation is based on a non-linear static analysis. The computation is done by the software MECANO. The initial meshing position gives a slight interpenetration of the two teeth in contact. It is the starting point of the iterative research of the static equilibrium of the system. The contact area is updated during the computation. It moves and spreads until the equilibrium is reached. The result is the starting point of the iterations for the next angular position of the driven gear. The algorithm solves the contact problem for each given meshing positions.

4.2. Application case

Data defining the example considered here are presented in the first six lines of Table 1. The second part of this table gives the values of the six coefficients of the modified roll method obtained after a manual optimization process. These six values are the starting point of the automated contact path improvement process.

Table 1
Initial manufacturing data.

<table>
<thead>
<tr>
<th>Workpiece</th>
<th>Pinion</th>
<th>Gear</th>
</tr>
</thead>
<tbody>
<tr>
<td>Tooth number</td>
<td>23</td>
<td>30</td>
</tr>
<tr>
<td>Module</td>
<td>5</td>
<td>5</td>
</tr>
<tr>
<td>Whole depth</td>
<td>10.24</td>
<td>10.24</td>
</tr>
<tr>
<td>Addendum</td>
<td>5.6</td>
<td>3.65</td>
</tr>
<tr>
<td>Face angle</td>
<td>40.2</td>
<td>54.28</td>
</tr>
<tr>
<td>Pitch angle</td>
<td>37.483</td>
<td>52.517</td>
</tr>
<tr>
<td>MR coefficient 1</td>
<td>0.639404</td>
<td>0.794488</td>
</tr>
<tr>
<td>MR coefficient 2</td>
<td>¥0.003814</td>
<td>0</td>
</tr>
<tr>
<td>MR coefficient 3</td>
<td>0.000598</td>
<td>0.001704</td>
</tr>
<tr>
<td>MR coefficient 4</td>
<td>¥0.0009997</td>
<td>0</td>
</tr>
<tr>
<td>MR coefficient 5</td>
<td>0.000023</td>
<td>¥0.000024</td>
</tr>
<tr>
<td>MR coefficient 6</td>
<td>¥0.000168</td>
<td>0</td>
</tr>
</tbody>
</table>

The components of the relative displacement of the gear axes are deduced from a loaded FE analysis of the complete gearbox. They are given in Table 2.

Table 2
Displacement compounds of the gear axes.

<table>
<thead>
<tr>
<th>Component</th>
<th>Value</th>
</tr>
</thead>
<tbody>
<tr>
<td>Pinion axial displacement</td>
<td>0.447472</td>
</tr>
<tr>
<td>Gear axial displacement</td>
<td>¥0.065310</td>
</tr>
<tr>
<td>Offset displacement</td>
<td>¥0.401187</td>
</tr>
<tr>
<td>Shaft angle displacement</td>
<td>0.005277</td>
</tr>
</tbody>
</table>

The optimization process runs on a PC with 2 CPU clocked at 1800 MHz. It is stopped after about 7 h. No error is generated during the execution of the various modules, although a wide variety of configurations is considered. This confirms the robustness of the algorithms that have been developed [20].

The process progression is illustrated in Fig. 7. The convergence discontinuity observed results from the surface discretization. The pressure peaks are due to contacts on flank edges. Nevertheless, the overall trend shows a reduction of the maximum contact pressure. The gear specialist chooses the combination of values of variables which gives the best result. The optimal design is achieved here at the 21st iteration. Table 3 lists the related variables.

Fig. 5. New spiral bevel gear topography optimization process.

Fig. 6. Simplified model and boundary conditions.

Fig. 7. Progression of the contact pressure minimization process.
<table>
<thead>
<tr>
<th>Piece</th>
<th>Pinion</th>
</tr>
</thead>
<tbody>
<tr>
<td>Modified roll coefficient 1</td>
<td>0.640054</td>
</tr>
<tr>
<td>Modified roll coefficient 2</td>
<td>0.0005219</td>
</tr>
<tr>
<td>Modified roll coefficient 3</td>
<td>0.0001244</td>
</tr>
<tr>
<td>Modified roll coefficient 4</td>
<td>0.0000292</td>
</tr>
<tr>
<td>Modified roll coefficient 5</td>
<td>0.0001435</td>
</tr>
<tr>
<td>Modified roll coefficient 6</td>
<td>0.000445</td>
</tr>
</tbody>
</table>

5. Conclusion

An optimization process able to design in an automated way the shape of spiral bevel gear flanks has been presented. It leads to a significant reduction of the development time, while allowing strengthening of the quality of contact patterns by the reduction of the contact pressure. Its extension to the minimization of tooth contact errors seems possible, in order to contribute to the reduction of noise and vibration levels and therefore a higher durability of helicopter gearboxes.

References


Fig. 8. Model mesh and contact areas.

![Fig. 9. Load sharing before and after optimization.](image)

Fig. 9. Load sharing before and after optimization.

Fig. 10. Contact path before and after optimization.